

Fact Finding Report

Subject / Title:

Special Inspection

Project: **POLO ENERGETICO ZARATE 23**
 Shop Order No: **MB000169**
 Event ID: **NA**
 Period of Time: **2024-09-02 to 2024-09-05**

CT-2024-1425

Reference

Berlin	2024-09-02
Place	Date

Mehdi Taghrar

Author(s)

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Content Approved	Released

Protection Class: **Restricted**

AL: **N** ECCN: **N**

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Brief Summary:

The gas turbine MB000169 has been preserved and kept in storage at Hamburg Harbour since 2016. Before its delivery, the conservation procedures were refreshed to guarantee optimal preservation. As part of this process, a boroscope inspection was performed on September 3rd, 2024, to evaluate and document the unit's condition. The inspection concluded that the turbine remains in good condition.

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1. Summary / Introduction

The gas turbine MB000169 has been preserved and kept in storage at Hamburg Harbour since 2016. Before its delivery, the conservation procedures were refreshed to guarantee optimal preservation.

As part of this process, a boroscope inspection was performed on September 3rd, 2024, to evaluate and document the unit's condition. The inspection concluded that the turbine remains in good condition.

Findings:

1. The turbine exit casing (TEC) exhibited signs of surface corrosion in several areas.
2. Two of the half joint bolts showed signs of paint loss and corrosion.
3. The outer surface of the compressor vane carriers showed signs of corrosion. The pictures were taken from the access points to compressor stages #5, #9, and #13.
4. The outer ring of VLe5 and the inner ring of VLe13 show signs of corrosion.
5. The lower outer manhole cover and the lower inner manhole connecting elements were found with signs of corrosion.
6. The burner supports were observed to have corrosion spots on their surfaces, particularly at the contact points with the casing and on the connecting elements.
7. The gas passages of three premix burners (burners #10, #12, and #14) were examined, leading to the following observations:
 - The burners displayed brownish/reddish deposits in the vane holes.
 - Slight corrosion was noted in the ring channel.
 - Very small droplets of water were found in the inner channel of the burner #10.
8. The seal strip 1 between TVC and casing 3 was found with signs of corrosion.

Recommendation:

- 1.-8. The findings can be left as found.

2. Personnel

2.1. Siemens Energy Personnel

Name	Job Description	Shift	Date In	Date Out
Mehdi Taghrrar	CT Fact Finder	Day Shift	2024-09-03	2024-09-06

3. Unit Information

3.1. Unit Data

Turbine S.O. No	MB000169	Generator S.O. No	EG000228
Turbine Frame	SGT5-4000F(8)	Generator Frame	SGEN5-1200A 118/55
Rated Speed		Unit Configuration	
Fuel(s)	Dual Fuel	I&C-System	
Turboset was shut down at		Turboset was shut down on	

4. Findings / Recommendations

Insp. Item 1 - Outer overall condition

Insp. Item 1.1 - Compressor section

Design group inspection according to Siemens Energy OEM guidelines indicates no findings.

Finding

The Compressor casing Surface, the IGV and the IGV support roller were found in good condition, no sign of corrosion.

Photo Attachments



RHS overview



RHS Compressor



RHS compressor (1)



LHS compressor side



RHS turbine side



RHS turbine side



LHS turbine side



RHS Burner part



IGV support roller



IGV support roller (1)



IGV ring



IGV levers

Insp. Item 1.2 - Turbine section

Design group inspection according to Siemens Energy OEM guidelines indicates following findings:

Finding

1. The turbine exit casing (TEC) exhibited signs of surface corrosion in several areas.
2. Two of the half joint bolts showed signs of paint loss and corrosion.

Recommendations

The findings can be left as found.

Photo Attachments



The turbine exit casing (TEC)



The turbine exit casing (TEC) (1)



The half joint bolt



The half joint bolt (1)



The half joint bolt (2)

Insp. Item 2 - Inner inspection

Insp. Item 2.1 - Accessibilities

Design group inspection according to Siemens Energy OEM guidelines indicates following findings:

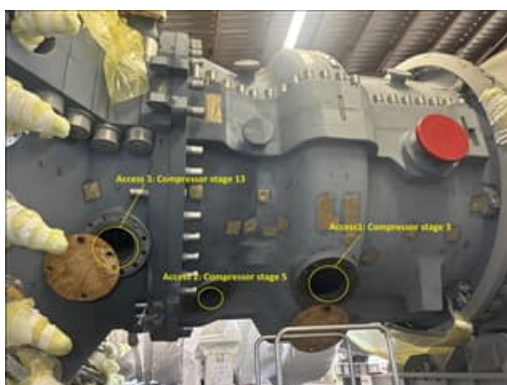
Finding

The compressor inspection was carried out through the openings at the compressor extraction points, as depicted in the images labeled "Accessibilities." These openings allowed for the examination of compressor stages #5, #9, and #13.

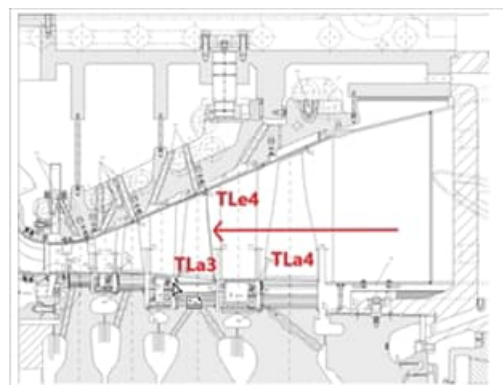
To inspect the gas passage of the premix burner, we accessed the gas pipe connection, as shown in the image titled "Premix Burner Gas Passage." A boroscope was used to inspect three premix burners: #10, #12, and #14.

Additionally, the turbine inspection was performed through the turbine exit casing, illustrated in the image labeled "Turbine Accessibilities."

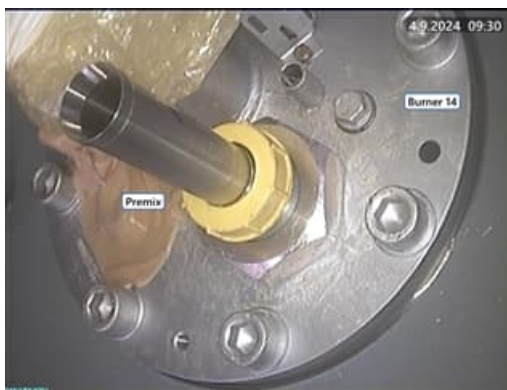
Photo Attachments



Accessibilities



Turbine accessibilities



Burner #14 overview



Premix gas passage burner #12

Insp. Item 2.2 - Compressor

Design group inspection according to Siemens Energy OEM guidelines indicates following findings:

Insp. Item 2.2.1 - Compressor vane carriers

Finding

The outer surface of the compressor vane carriers showed signs of corrosion. The pictures were taken from the access points to compressor stages #5, #9, and #13.

Recommendations

The finding can be left as found.

Photo Attachments



2409040054



2409040056



IMG_4927



IMG_4935



IMG_4959



IMG_4966

Insp. Item 2.2.2 - IGV and compressor inlet

Finding

The compressor inlet area and the IGV region were assessed to be in good condition, with no indications of corrosion observed.

Recommendations

The findings can be left as they are, with ongoing monitoring to be conducted during future inspection.

Photo Attachments



IMG_4812



IMG_4814



IMG_4816



IMG_4829



IMG_4835



IMG_4845

Insp. Item 2.2.3 - Compressor vanes

Finding

The compressor inlet area and the IGV region were assessed to be in good condition, with no indications of corrosion observed.

Recommendations

The findings can be left as they are, with ongoing monitoring to be conducted during future inspection.

Photo Attachments



Compressor stage #9 (1)



Compressor stage #9 (2)



Compressor stage #9 (3)



Compressor stage #5



Compressor stage #5 (1)



Compressor stage #5 (2)



Compressor stage #5 (3)



Compressor stage #13.

Insp. Item 2.3 - Combustion system

Design group inspection according to Siemens Energy OEM guidelines indicates following findings:

Insp. Item 2.3.1 - The lower inner and outer manhole

Finding

The lower outer manhole cover and the lower inner manhole connecting elements were found with a sign of corrosion.

Recommendations

The finding can be left as found.

Photo Attachments



The lower inner manhole



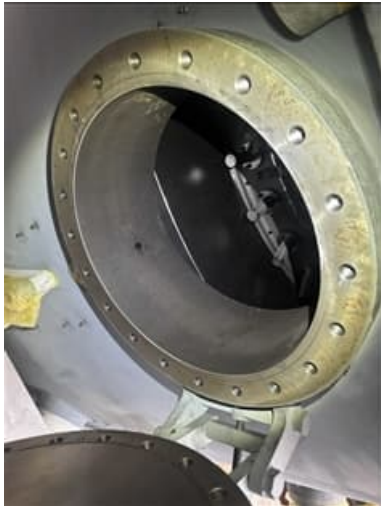
The lower inner manhole (1)



The lower outer manhole



The lower outer manhole (1)



The lower outer manhole (2)



The lower outer manhole (3)

Insp. Item 2.3.2 - Burner supports

Finding

The burner supports were observed to have corrosion spots on their surfaces, particularly at the contact points with the casing and on the connecting elements.

Recommendations

The finding can be left as found.

Photo Attachments



Burner overview



Burner support



Burner support (1)



Burner support (2)



Burner support (3)

Insp. Item 2.3.3 - Premix burners gas passage and inner channel

Finding

The gas passages of three premix burners (positions #10, #12 and #14) were examined, leading to the following observations:

1. The burners displayed brownish/reddish deposits in the vane holes.
2. Slight corrosion was noted in the ring channel.
3. Very small droplets of water were found in inner channel of the burner #10.

Recommendations

The finding can be left as found.

Photo Attachments



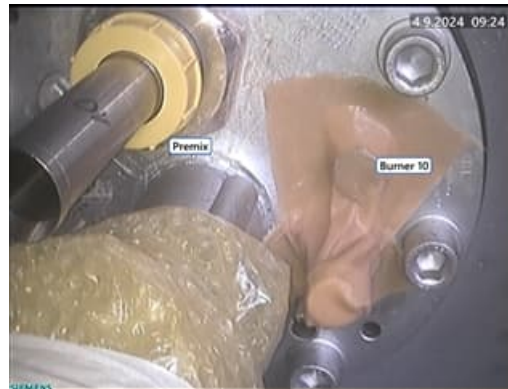
Burner #12 Premix gas passage



Burner #12 inner Chanel



Burner #12 inner Chanel (1)



Burner #10



Burner #10 Premix gas passage



Burner #10 inner Chanel



Burner #10 inner Chanel (1)



Burner #10 inner chanel with droplets of water



Burner #14 Premix gas passage



Burner #14 inner Chanel



Burner #14 inner Chanel (1)



Burner #14 inner Chanel (2)

Insp. Item 2.4 - Turbine side

Design group inspection according to Siemens Energy OEM guidelines indicates no findings.

Insp. Item 2.4.1 - Turbine blading

Photo Attachments



T1a4 overview



TLa 4



TLa4 (1)



TLa4 (2)



2409040209



2409040214



2409040218



2409040219